

Original Research Paper

Analysis of Combined Natural Convection and Radiation Heat Transfer Using a Similarity Solution

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Abstract: A similarity solution for laminar boundary layer in a buoyancy induced flow over an isothermal, vertical wall in the presence of radiation is developed. Radiation heat transfer is taken into account by application of the Rosseland approximation in the energy conservation equation. The family of similarity solutions to the governing equations is presented. The computed results show that thermal and momentum boundary layer thickness values are increased as a result of radiation heat transfer. Furthermore for higher values of the radiation parameter, the maximum velocity inside the boundary layer increases.

Keywords: Natural Convection, Combined Radiation-Convection, Similarity Solution, Boundary Layer

Introduction

Evaluating the heat transfer interactions involving combined convection and radiation is important in many engineering applications, especially at high temperatures. Natural convection flow analysis has been the main subject of numerous researches due to its many applications in engineering and industrial processes. At high temperatures, when flowing substance is hot enough to emit and absorb radiant energy we are dealing with a combined radiation-convection heat transfer mode. Radiation effect on flow is most significant in situations where, convection heat transfer is limited; hence radiation could have a high impact on free convection instances of absorbing-emitting fluids (Cess, 1966). Energy transfer through participating media has received a lot of attention during past years. Some recent important applications are high temperature solar energy receivers, high temperature thermal storage, advanced power generation systems, radiant furnaces, molten glass containers and so on.

The receiver of a Concentrated Solar Power (CSP) plant represents an example where combined radiation-convection heat transfer could be expected. The Solar receiver of a CSP plant is the heat exchanger where the concentrated solar radiation is absorbed and converted into heat. Temperature of the solar receiver can reach more than 700 °C in a solar tower. Precise evaluation of heat losses from the structure due to convection and radiation is a complex problem. The convective and radiative heat transfer between the surface and the hot fluid adjacent to it are coupled and none of the heat

transfer modes could be evaluated separately. The presence of radiative heat transfer affects the velocity and temperature profiles inside the momentum and thermal boundary layers and alters the overall heat transfer coefficient.

Although some of these applications are relatively recent, the importance of this subject has been recognized since the 1960s. Cess (1966) analyzed the interaction of thermal radiation with free convection for a gray gas in the optically thick region. He solved the problem by means of singular perturbation technique for small values of radiation parameter ($N_R \ll 1$). Arpaci (1968) investigated the effect of thermal radiation on local heat flux for laminar free convection from a heated vertical plate. He approximated the radiation heat flux by an integro exponential function and selected first order polynomial to approximate the velocity and temperature profiles. Due to lengthy expressions, no results were presented for temperature and velocity profiles and only local heat flux for thick and thin gas assumptions was presented. Cheng and Özişik (1972) considered the heat transfer for both hot and cold wall by simultaneous radiation and free convection in an absorbing, emitting, isotropically scattering gray fluid by solving non-similar momentum and energy equations. The radiant part of the problem was treated by the application of a normal-mode-expansion technique. Their results show that both temperature and velocity boundary layers get thicker by decreasing radiation parameter (N_R). Hossain and Takhar (1996) have studied the effects of radiation on forced and free convection, boundary layer flow along a vertical flat

plate with uniform free stream and surface temperatures. They used the Rosseland diffusion approximation for radiation, which leads to non-similarity solutions for the flow. In their findings, they declared that both dimensionless local shear stress and local heat transfer rate decline when N_R increases. In another study, Hossain *et al.* (1999) studied the same problem for a porous vertical plate. Molla *et al.* (2011) investigated the effect of radiation on the natural convection flow of an optically thick viscous incompressible fluid along a vertical plate with sinusoidal temperature distribution. Siddiqa *et al.* (2014) investigated the effects of radiation heat transfer on the natural convection boundary layer flow over a wavy horizontal surface.

Bataller (2008a; 2008b) analyzed the effects of thermal radiation on the laminar boundary layer adjacent to a flat plate in a uniform stream of fluid (Blasius flow) and also on a moving plate in a quiescent ambient fluid (Sakiadis flow) under convective-radiative surface boundary conditions. Also, many studies on laminar and turbulent natural convection heat transfer coupled with radiation heat transfer in the presence of a radiative surface and/or participating medium in an enclosure have been published (Xamán *et al.*, 2008; Moghadassian and Kowsary, 2014; Bouali *et al.*, 2006). Thermal radiation effect in natural convection for a cylinder has been investigated by a number of researchers. For example, Novotny and Kelleher (1967) investigated the laminar free convection of an absorbing, emitting gas in the region of the stagnation point of a horizontal cylinder. The radiation effect on free convection along an isothermal vertical cylinder and mixed convection from a horizontal cylinder were studied by Hossain *et al.* (1998; Hossain and Alim, 1997; Hossain *et al.*, 1999).

Including radiation heat transfer mode into energy equation, leads to a nonlinear partial differential equation. Almost all the previous works in this field utilized an approximate method to solve this differential or in some cases integro-differential equation. As a result, any parametric analysis of the velocity and temperature profiles was lengthy and difficult.

The main objective of the present work is to investigate the effects of radiation heat transfer on momentum and heat transfer boundary layers. The velocity and temperature distribution profiles inside the boundary layer for an optically thick, incompressible, natural buoyancy induced flow past a hot vertical wall with constant temperature are evaluated. By using Rosseland diffusion approximation and making a series of transformations on governing equations, we delivered the similarity solution for the flow field. This procedure simplifies the analysis and makes interpretation of the results easier.

Governing Equations

Consider laminar boundary layer flow over a hot flat plate with constant temperature, which is driven by the buoyancy forces depicted in Fig. 1. To simplify the problem, the following assumptions are made:

- Steady-state two-dimensional conditions in which the gravity force acts in the negative x direction
- General Boussinesq assumption and taking into account the boundary layer approximation
- Gray fluid, emitting and absorbing radiation with no scattering effect
- Constant physical properties
- Viscous dissipation has been neglected in the energy, due to the small velocities associated with the free convection

The governing equations take the form of (Equations (1) to (3)):

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0 \quad (1)$$

$$u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} = \nu \frac{\partial^2 u}{\partial y^2} + g\beta(T - T_\infty) \quad (2)$$

$$u \frac{\partial T}{\partial x} + v \frac{\partial T}{\partial y} = \frac{k}{\rho c_p} \frac{\partial^2 T}{\partial y^2} - \frac{1}{\rho c_p} \frac{\partial q_r}{\partial y} \quad (3)$$

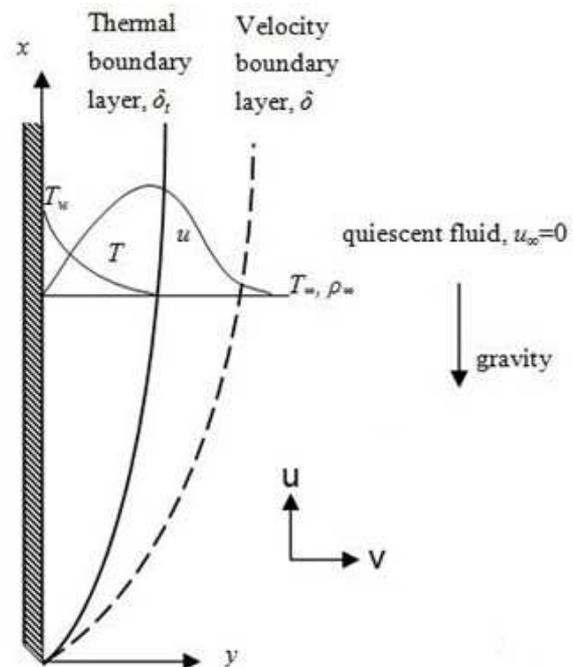


Fig. 1. Momentum and thermal boundary layer development on a hot vertical wall

Here u and v are velocity components in x and y directions, g is the acceleration due to gravity, β is the coefficient of thermal expansion, ν is the kinematic viscosity, T is the temperature, k is the fluid conductivity, ρ is the fluid density and c_p is the heat capacity of the fluid.

The boundary conditions (Equation 4) for the velocity field are:

$$\begin{aligned} u = v = 0 \quad \text{at } y = 0 \\ u = 0 \quad \text{at } x = 0 \\ u \rightarrow 0 \quad \text{as } y \rightarrow \infty \end{aligned} \quad (4)$$

The boundary conditions (Equation 5) for the temperature field are:

$$\begin{aligned} T = T_w \quad \text{at } y = 0 \\ T = T_\infty \quad \text{as } y \rightarrow \infty \end{aligned} \quad (5)$$

Radiation heat transfer is added in the energy equation as one dimensional heat flux in y direction. Radiative heat fluxes can be approximated by the Rosseland diffusion approximation (Rosseland, 1936) for an optically dense medium, which has been used in many radiation related studies (Cess, 1966; Arpacı, 1968; Cheng and Özişik, 1972; Hossain and Takhar, 1996; Hossain *et al.*, 1999; Molla *et al.*, 2011; Siddiqa *et al.*, 2014; Bataller, 2008a; 2008b) and (Hossain *et al.*, 1998; Hossain and Alim, 1997; Hossain *et al.*, 1999). Using the Rosseland diffusion approximation for radiation, the radiation heat flux is simplified as:

$$q_r = -\frac{4\sigma}{3\alpha_R} \frac{\partial T^4}{\partial y} \quad (6)$$

where, σ and α_R are the Stefan-Boltzmann constant and the Roseland mean absorption coefficient, respectively. It is possible to assume the temperature profile within the flow such as that, the term T^4 be expressed as a linear function of temperature. Hence, expanding T^4 in a Taylor series about T_∞ and neglecting higher-order terms we get:

$$T^4 \cong 4T_\infty^3 T - 3T_\infty^4 \quad (7)$$

Using Equations (6) and (7), the energy Equation (3) becomes:

$$u \frac{\partial T}{\partial x} + v \frac{\partial T}{\partial y} = \left(\alpha + \frac{16\sigma T_\infty^3}{3\rho c_p \alpha_R} \right) \frac{\partial^2 T}{\partial y^2} \quad (8)$$

Here α is the fluid thermal diffusivity. As it can be seen in equation (8), the effect of radiation is manifested in the form of enhanced thermal diffusivity.

By introducing radiation parameter as $N_R = \frac{ka_R}{4\sigma T_\infty^3}$

Equation (8) can be rearranged as (Equation (9)):

$$u \frac{\partial T}{\partial x} + v \frac{\partial T}{\partial y} = \frac{\alpha}{k_0} \frac{\partial^2 T}{\partial y^2} \quad (9)$$

where, $k_0 = \frac{3N_R}{3N_R + 4}$. The radiation parameter (N_R) is the

parameter that measures the importance of conduction versus radiation within the fluid (Cheng and Özişik, 1972; Sparrow and Cess, 1966). It can be seen that, at $k_0 = 1$, the effect of thermal radiation is not taken into account. Table 1 shows the range of N_R for carbon dioxide, ammonia vapor and water vapor. Now by defining a similarity variable it is possible to use the similarity solution to determine the velocity and temperature profiles adjacent to the plate.

Similarity Variable Method

The idea behind the similarity solution is that the velocity and temperature profiles in different x location in the boundary layer are geometrically similar, differing only by a stretching factor in the x direction (Kays *et al.*, 2005). Similarity solutions to the laminar boundary layer equations for steady natural convection from an isothermal vertical flat plate have been known since the pioneering work of Schmidt and Beckmann (1930). Sparrow and Gregg (1958) added solutions for non-isothermal vertical plates with surface temperatures of the form $T_w - T_\infty = Ax^n$ and Be^{mx} .

In order to derive a general solution, the following forms for similarity variable (η) (Equation (10)), stream function (ψ) (Equation (11)) and dimensionless temperature (θ) (Equation (12)), are considered:

$$\eta = \frac{y}{x} \left(\frac{1}{4} Gr_x \right)^{1/4} \quad (10)$$

$$\psi(x, \eta) = 4\nu F(\eta) \left(\frac{1}{4} Gr_x \right)^{1/4} \quad (11)$$

$$\theta = \frac{T - T_\infty}{T_w - T_\infty} = \theta(\eta) \quad (12)$$

where, F is the velocity function and Gr_x is the local Grashof number defined as (equation (13)):

$$Gr_x = \frac{g\beta(T_w - T_\infty)x^3}{\nu^2} \quad (13)$$

The velocity components in the x (Equation (14)) and y (Equation (15)) directions can be expressed in terms of similarity variables by the following equations:

$$u = \frac{\partial \psi}{\partial y} = \frac{2\nu}{x} Gr_x^{1/2} F'(\eta) \quad (14)$$

$$v = -\frac{\partial \psi}{\partial x} = \frac{\nu}{x} \left(\frac{Gr_x}{4} \right)^{1/4} [\eta F'(\eta) - 3F(\eta)] \quad (15)$$

Here the prime denotes differentiation with respect to η . The stream function has been introduced in a way that automatically satisfies the continuity equation. The momentum (Equation (2)) and energy (Equation (9)) equations, using the above transformations reduce to (Equation (16) and Equation (17)):

$$F''' + \theta + 3FF'' - 2(F')^2 = 0 \quad (16)$$

$$\theta' + 3k_0 Pr F \theta' = 0 \quad (17)$$

The boundary conditions transform to (equation (18) and equation (19)):

$$F(\eta) = F'(\eta) = 0, \quad \theta(\eta) = 1 \text{ at } \eta = 0 \quad (18)$$

$$F'(\eta) = 0, \quad \theta(\eta) = 0 \text{ as } \eta \rightarrow \infty \quad (19)$$

Without considering the radiation, the Prandtl number appears explicitly in Equation (17) and the solutions are expected to be of the form $F=F(\eta, Pr)$ and $\theta = \theta(\eta, Pr)$. However, when radiation effects are considered, another parameter must be taken into account, which is the radiation parameter. It is possible to present the effect of radiation heat transfer by introducing the modified Prandtl number as $Pr^* = k_0 Pr$. Solutions for different Prandtl numbers are available in many references (Sparrow and Cess, 1966; Schuh, 1948; Ostrach, 1953; Webb, 1990) and it is possible to exploit desirable results for modified Prandtl number.

Numerical Solution and Results

In order to visualize the effect of radiation heat transfer, in Table 2, the values of Pr^*/Pr for different values of N_R are presented. By increasing the value of N_R the ratio of the convection heat transfer rate to radiation heat transfer rate increases. For small values of radiation parameter, radiation is the dominating heat transfer mode. By increasing the value of N_R , ratio of the convection heat transfer to radiation heat transfer increases. As it is shown in Table 2 the values of modified Prandtl number to conventional Prandtl number is always less than one ($k_0 < 1$). With the increasing of radiation parameter this ratio converges to one and for N_R more than 50, this ratio remains close to one. It is obvious as N_R increases, the value of k_0 becomes closer to unity and the effect of radiation becomes less considerable on the flow.

Equations (16) and (17) are coupled nonlinear ordinary differential equations. Therefore any changes in parameters Pr^* will affect both fluid velocity and its temperature distribution in the boundary layer.

Table 1. Different gases and their N_R values (Ali *et al.*, 1984)

	T [°F]	Pr	NR
Carbon dioxide	100-650	0.76-0.60	10-30
Ammonia vapor	120-400	0.88-0.84	30-200
Water vapor	220-900	1	30-200

Table 2. Different gases and their N_R values (Ali *et al.*, 1984)

NR	Pr^*/Pr
0.01	0.0074
0.1	0.0700
0.5	0.2720
1	0.4290
5	0.7900
15	0.9180
25	0.9380
50	0.9740
100	0.9870
1000	0.9990

Equation (17) is a second order ordinary differential equation and can be broken down into two first order ordinary differential equations (equation (20) and equation (21)):

$$\theta' = A \quad (20)$$

$$A' = -3Pr^* F \theta' = g(F, A) \quad (21)$$

Equation (16) is a third order ordinary differential equation and can be broken down into three first order ordinary differential equations (equations (22) to (24)):

$$F' = B \quad (22)$$

$$B' = C \quad (23)$$

$$C' = 2B^2 - 3FC - \theta = f(B, F, C, \theta) \quad (24)$$

By use of fourth order Runge-Kutta algorithm along with shooting procedure these equations are solved simultaneously. There are three boundary conditions at $\eta = 0$ which are: $F(0) = F'(0) = 0$ and $\theta(0) = 1$ and two boundary conditions at $\eta \rightarrow \infty$ which are: $F'(\infty) = \theta(\infty) = 0$. We guess the values of $F''(0)$ and $\theta'(0)$ and compare the calculated values at the other end of the domain ($\eta \rightarrow \infty$) with the existing boundary conditions. This procedure continues until the desirable convergence between calculated values and boundary values are reached. The problem was solved with $\Delta_\eta = 0.01$ and a complete determination of the solution has been carried out.

There are very limited experimental studies available on effect of combined radiation-convection heat transfer inside boundary layer in literature. Webb (1990) conducted an experimental and analytical study to explore the interaction between laminar free convection and radiative heat transfer from an isolated vertical plate with isoflux heating. Experiments were formulated and carried out in air environment at moderate pressure, therefore negligible volumetric radiation-absorption was

considered inside the boundary layer. A similar study by Sabareesh *et al.* (2010) was carried out for a hot plate with isoflux and isothermal boundary conditions. In an experimental study the effects of gas-gas radiation on thermal fields in a gas flowing by free convection around a finite isothermal plate was investigated by Lacona and Taine (2001). Holographic interferometry and laser beam deflection techniques were used to show the radiation heat transfer effects on boundary layer for both absorbing (CO₂) and non-absorbing (N₂) mediums. Although temperature distribution perpendicular to the plate and local heat transfer coefficient along the plate were presented for non-absorbing medium, for absorbing-emitting medium only the heat transfer coefficient is reported.

To compare the modified Prandtl number with experimental results following equations are used for calculating the local heat transfer coefficient along the plate (equation (25)):

$$h_x = \frac{k}{x} \left(-\frac{\theta'(0)}{\sqrt{2}} \right) Gr_x^{1/4} \quad (25)$$

Here Gr_x is the local Grashof number and k is thermal conductivity of the medium. The thermo-physical properties of the medium (CO₂) is calculated at film temperature (equation (26)):

$$T_f = \frac{T_w + T_\infty}{2} \quad (26)$$

where, T_w is the wall temperature and T_∞ is considered as the cavity temperature equal to temperature of undisturbed fluid. Figure 2-4 show the calculated heat transfer coefficient and experimental data from (Lacona and Taine, 2001). Three different conditions for T_w and T_∞ were analyzed.

As mentioned by Lacona and Taine (2001; Gebhart *et al.*, 1988), the gas has been preheated by convection along horizontal edge of the plate and radiation from cavity walls; also boundary layer thickness does not change considerably in the range of 10-40 mm and approximately corresponds to the difference between theoretical and experimental values. Also the optically thick approximation is not completely valid at the leading edge of the plate, which makes the analytical results lower the experimental data.

The average difference between experimental and analytical results is 21.5%.

To examine the outcome of modified Prandtl method with other analytical methods, comparisons have been made with results presented by Hossain *et al.* (1999) and Cheng and Özişik (1972).

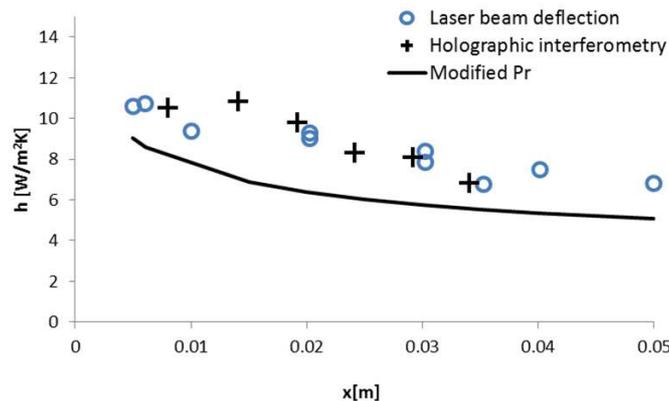


Fig. 2. Local heat transfer coefficient along the plate, T_w = 523 K, T_∞ = 473 K, N_R = 22, Pr = 0.74

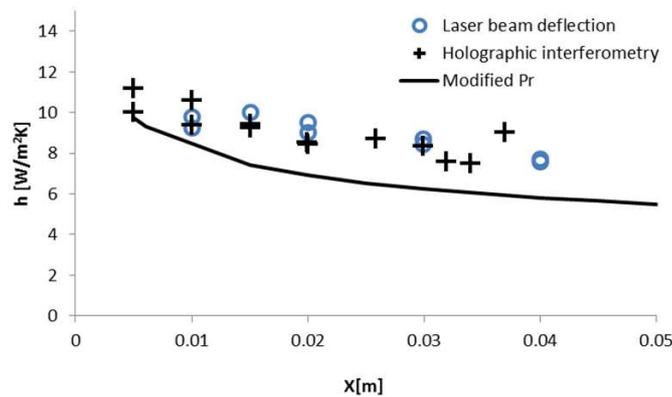


Fig. 3. Local heat transfer coefficient along the plate, T_w = 603 K, T_w = 543 K, N_R = 27, Pr = 0.73

Figure 5 and 6 show the effect of radiation on velocity and temperature distribution for a fluid with $Pr=1$ and $N_R = 1.0$. The results using modified Prandtl number are very close to the results presented by Hossain *et al.* (1999).

Figure 7 and 8 show the effect of radiation on boundary layer velocity and temperature distributions for a fluid with $Pr = 0.733$ and $N_R = 1.0$. The results for temperature distribution using modified Prandtl number are very close to the results presented by Cheng and Özişik (1972), but velocity profile which is presented by Cheng and Özişik (1972) is not for optically thick gas and therefore some mismatch is expected.

As it is depicted in Figures 5 through 8, the maximum flow velocity inside the boundary layer is

up to 26.3% higher for $Pr = 1.0$ and up to 24.4% higher for $Pr = 0.733$, when radiation heat transfer is considered compared to no radiation case. For fluid with larger value of Pr number, the effect of radiation heat transfer inside the momentum boundary layer is more prominent with respect to the fluid with a lower Pr number.

Furthermore, it is shown that the temperature distribution profile inside the boundary layer becomes more uniform due to presence of radiation heat transfer compared to no radiation case. In addition both momentum and thermal boundary layers get thicker when radiation heat transfer is present. Further details are presented in Fig. 9 and 10 and Table 3.

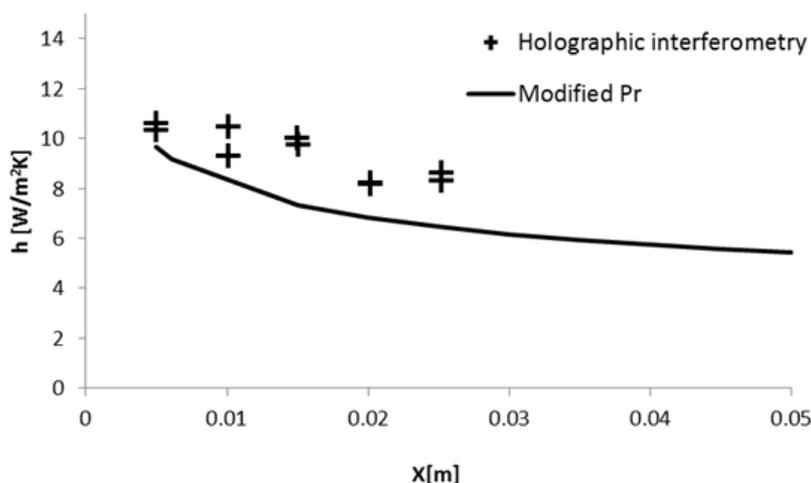


Fig. 4. Local heat transfer coefficient along the plate, $T_w = 643$ K, $T_\infty = 583$ K, $N_R = 29$, $Pr = 0.72$

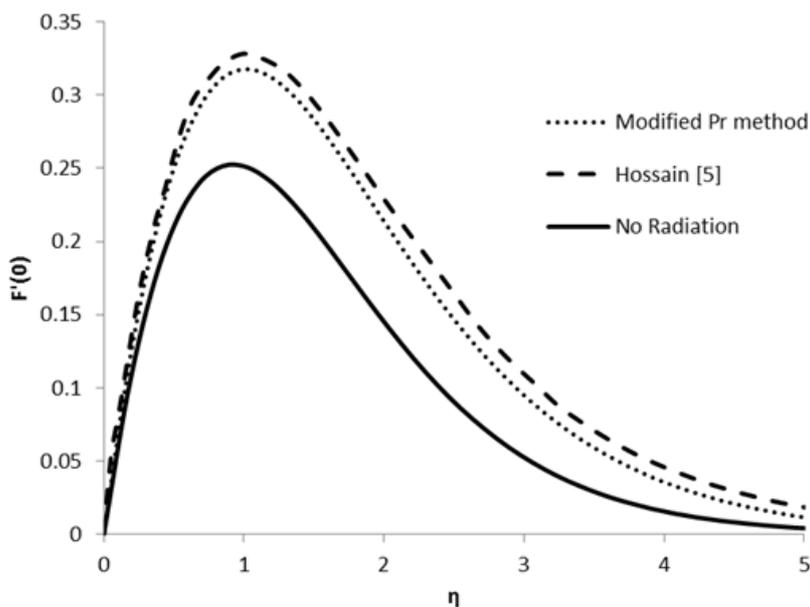


Fig. 5. Velocity distribution as a function of η , $N_R = 1.0$, $Pr = 1.0$

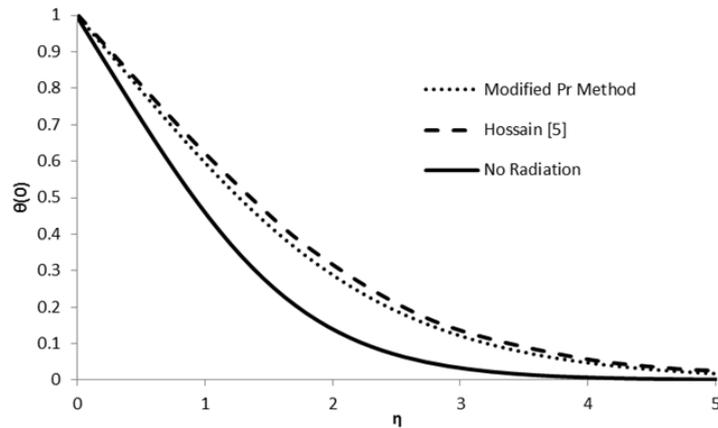


Fig. 6. Temperature distribution as a function of η , $N_R=1.0$, $Pr=1.0$

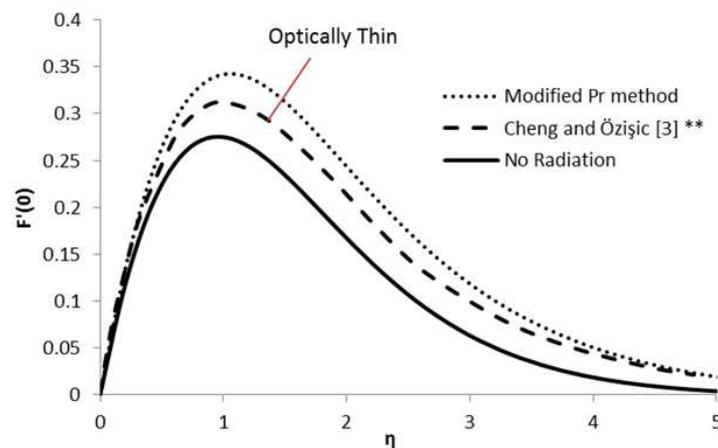


Fig. 7. Velocity distribution as a function of η , $N_R = 1.0$, $Pr = 0.733$

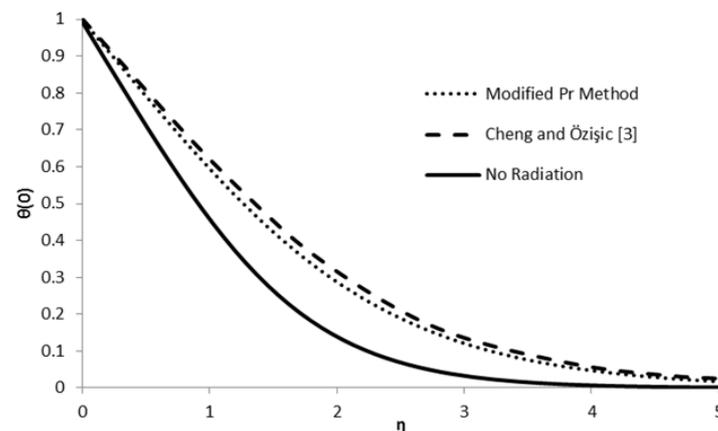


Fig. 8. Temperature distribution as a function of η , $N_R = 1.0$, $Pr = 0.733$

The results for distribution of $F'(\eta) \propto u$ and $\theta(\eta)$ in a range of N_R numbers between 0.5 and 10 are shown in Fig. 9 and 10, which are in the acceptable range in accordance with data presented in (Gebhart *et al.*, 1988) for no radiation condition and for different Prandtl numbers.

Numerical value of k_0 is always between zero and one, therefore the value of Pr^* will always be less than the value of Pr . Table 3 presents the enhancement of momentum boundary layer thickness and maximum velocity inside the boundary layer for different N_R numbers compared to no radiation condition.

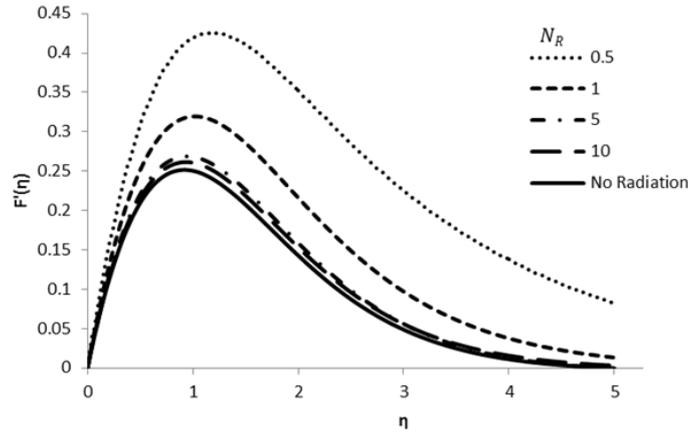


Fig. 9. Boundary region velocity distribution as a function of η for different N_R at $Pr = 1$

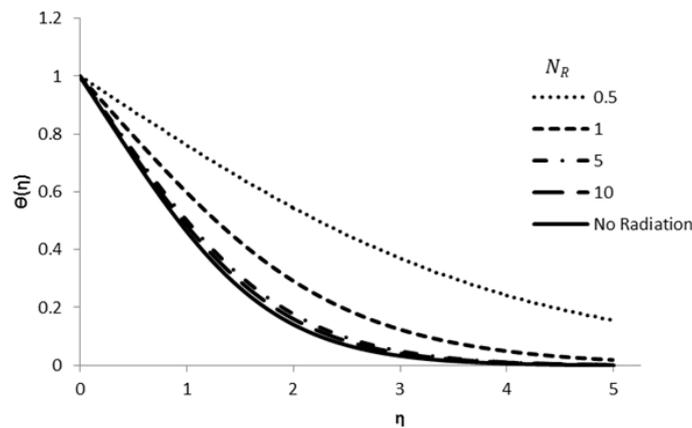


Fig. 10. Boundary region temperature distribution as a function of η for different N_R at $Pr = 1$

Table 3. Enhancement of momentum boundary layer thickness and maximum velocity in boundary layer for different NR, at $Pr = 1$

NR	Enhancement of boundary layer thickness	Enhancement of max. velocity inside the boundary layer
(no radiation)	1.00	1.00
10	1.01	1.03
5	1.04	1.07
1	1.20	1.26
0.5	1.46	1.41

As it is presented in the Table 3 by reduction of Pr^* (lower radiation parameter), thickness of momentum boundary layer increases and also the maximum value of velocity inside the boundary layer increases. Similar conclusions are reported by Luo *et al.* (2014). For N_R value equal to 0.5 the momentum boundary thickness is enhanced by 46% and maximum velocity inside the boundary layer increases by 41% at $Pr = 1$.

Conclusion

In this study, the effect of radiation on laminar natural convection flow adjacent to a vertical isothermal wall has been investigated. Unlike other

researches done in this area, the introduced analytical procedure provides an opportunity to incorporate the effects of radiation fast and with fewer amounts of calculations. Radiation heat transfer effect is taken into account by Rosseland approximation and considering the radiation parameter in governing energy balance equation. This parameter is a tool to determine the importance of radiation heat transfer in fluid flow inside the boundary layer. It is illustrated that as the value of radiation parameter increases, a diminution in the thermal radiation effect appears.

By manipulating the energy equation and using the similarity solution method, two ordinary differential equations have been established to determine the

velocity and temperature profiles inside the boundary layer. The only difference of these equations with conventional natural convection equations is a new parameter (k_0) which is multiplied to Prandtl number and resulting in a modified Prandtl number. Because the value of k_0 is always less than one, the value of modified Prandtl number is always less than the Prandtl number. Hence, the presence of radiative heat transfer results in increasing the thickness of the momentum boundary layer and also increasing the maximum fluid velocity inside the boundary layer.

Effect of radiation heat transfer on the velocity and temperature profiles in the boundary layer for most of gases is not considerable for $N_R \geq 10$ and can be ignored for N_R higher than 50. But for lower values of N_R , neglecting radiation heat transfer will result in significant errors in heat transfer coefficient and skin friction.

Author's Contributions

Mehdi Zeyghami: As the first and corresponding author discovered the new method of solving the problem, executed the numerical analysis and extracted the results and, contributed to the writing of the manuscript.

Muhammad M. Rahman: As the second author of the paper designed the research plan and organized the study and contributed to the writing of the manuscript

Ethics

We maintained highest possible ethical standards Integrity regarding data gathering and analysis

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Nomenclature

c_p	Specific heat at constant pressure [J/kgK]
F	Dimensionless stream function
g	Gravitational acceleration [m/s^2]
Gr_x	Local Grashof number
h_x	Local heat transfer coefficient [W/m^2K]
k	Thermal conductivity [W/mK]
N_R	Radiation parameter
Pr	Prandtl number
Pr^*	Modified Prandtl number
qr	Radiative heat flux [W/m^2]
T	Temperature [K]
T_w	Wall temperature [K]
T_∞	Ambient fluid temperature [K]
u	Velocity component in x-direction [m/s]
v	Velocity component in y-direction [m/s]
ν	Kinematic viscosity [kg/ms]
x	Coordinate along wall surface [m]
y	Coordinate normal to the wall surface [m]

Greek Symbols

α	Fluid thermal diffusivity [m^2/s]
α_R	Rosseland mean absorption coefficient
β	Coefficient of thermal expansion [$1/K$]
ρ	Density [kg/m^3]
σ	Stefan-Boltzmann constant [W/m^2K^4]
η	Similarity variable
ψ	Stream function
θ	Dimensionless temperature function
δ_t	Thermal boundary layer thickness [m]
δ	Velocity boundary layer thickness [m]